



**WAVELETS ON THE INTERVAL**

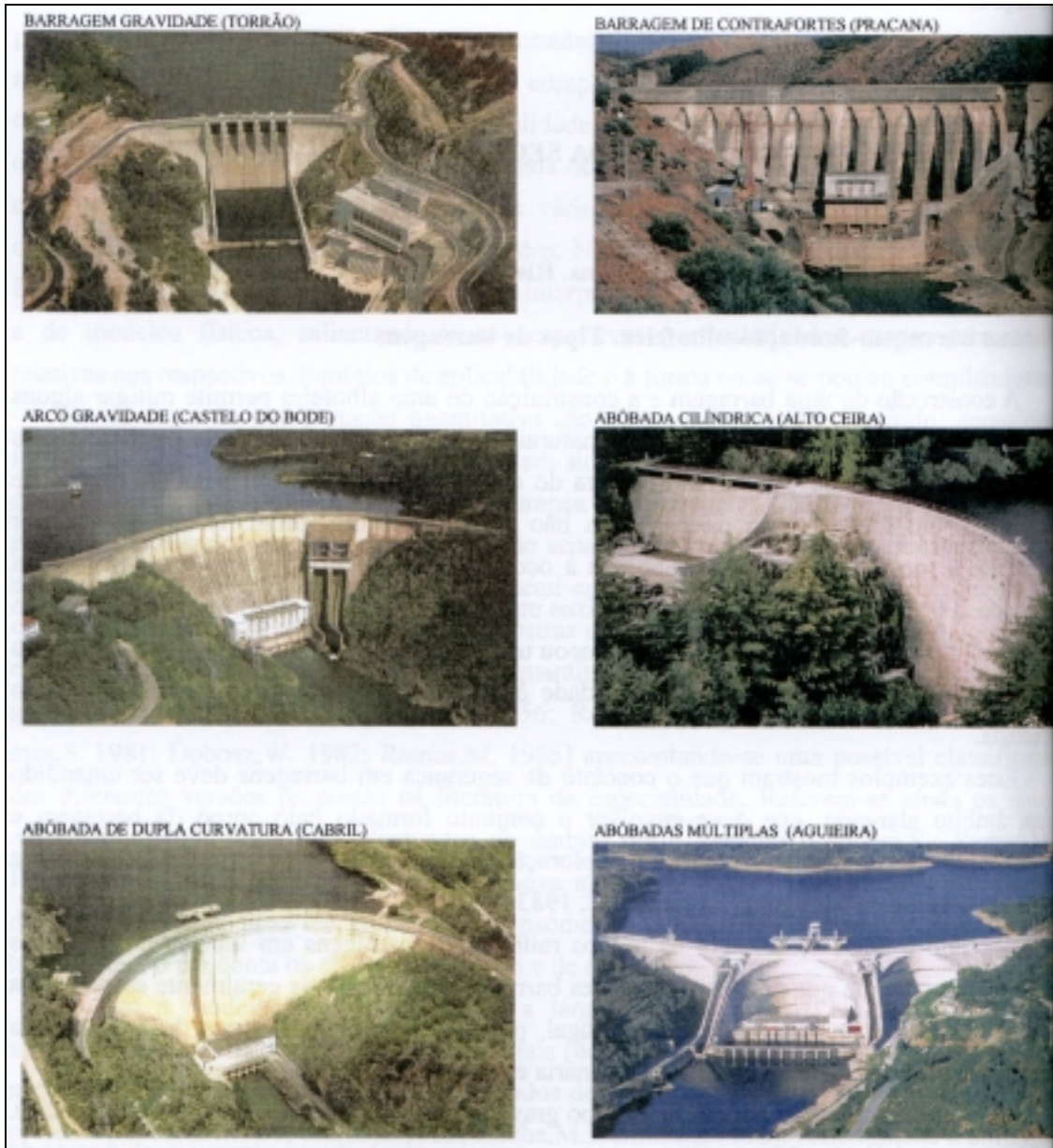
**APPLICATION TO ELASTICITY PROBLEMS**

9 –12 April 2001  
Marseille, France



- 3D
  - Dams (concrete)





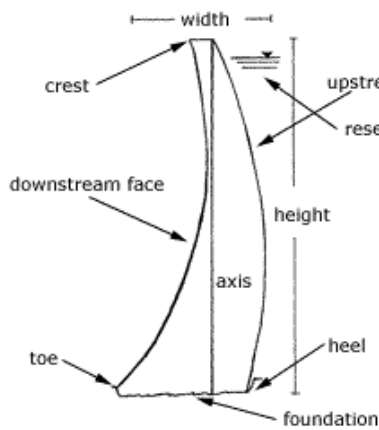
## ARCH DAMS



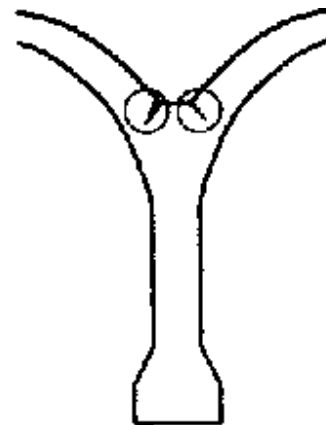
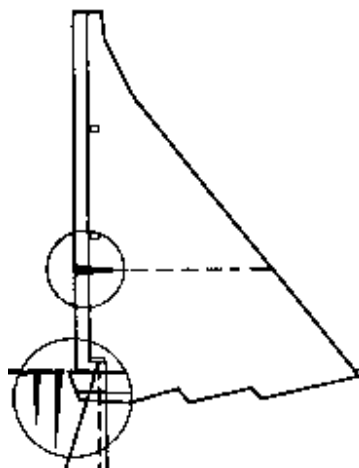
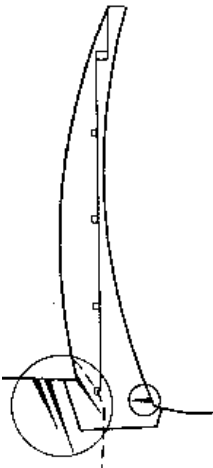
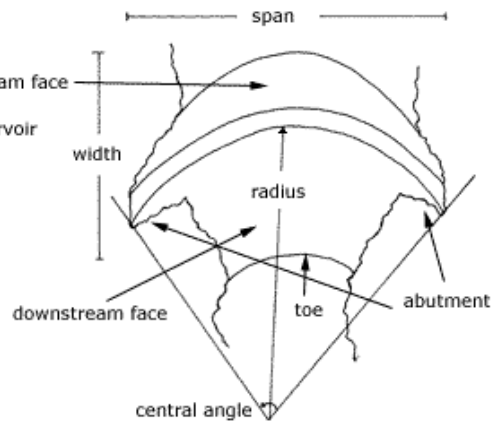
## BUTTRESS DAMS



cross section



plan view


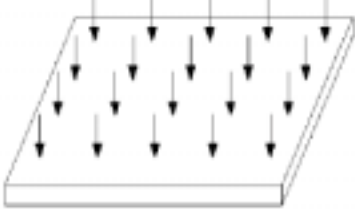


## 1. Introduction to the problems we study

- the domain  $V^e$
- the surface  $\Gamma$  divided in two complementary parts:
  - Dirichlet boundary  $\Gamma_u^e$
  - Neumann boundary  $\Gamma_\sigma^e$

	EQUILIBRIUM CONDITIONS	ELASTICITY CONDITIONS	COMPATIBILITY CONDITIONS
on $V^e$	$\sigma_{ij,i} + b_j = 0$ $\mathbf{D}\boldsymbol{\sigma} + \mathbf{b} = 0$	$\sigma_{ij} = -\frac{E}{1+\nu} \varepsilon_{ij} + \frac{E\nu}{(1+\nu)(1-2\nu)} \varepsilon_{kk} \delta_{ij}$ $\boldsymbol{\varepsilon} = \mathbf{f}\boldsymbol{\sigma}$	$\varepsilon_{ij} = \frac{1}{2} (u_{i,j} + u_{j,i})$ $\boldsymbol{\varepsilon} = \mathbf{D}^* \mathbf{u}$
on $\Gamma$	$\sigma_{ij} n_i = t_j$ $\mathbf{N}\boldsymbol{\sigma} = \mathbf{t}$		$u_i = \bar{u}_i$ $\mathbf{u} = \bar{\mathbf{u}}$



STRETCHING PLATES	BENDING PLATES
	

APPROXIMATION CRITERIA		
$\sigma = S \cdot X$	$u = U_V \cdot q_V$	$u = U_\Gamma \cdot q_\Gamma$

EQUILIBRIUM IN THE DOMAIN	EQUILIBRIUM ON THE BOUNDARY
$\int_V U_V^T (D\sigma + b) dV = 0$ $A_V = \int (DS)^T U_V dV$ $A_V^T X = -Q_V$	$\int_{\Gamma_\sigma} U_\Gamma^T (N\sigma - t) d\Gamma_\sigma = 0$ $A_\Gamma = \int_{\Gamma_\sigma} (NS)^T U_\Gamma d\Gamma_\sigma$ $A_\Gamma^T X = Q_\Gamma$



## COMPATIBILITY IN THE DOMAIN

$$\mathbf{e} = -\int_V (\mathbf{D}\mathbf{S})^T \mathbf{u} \cdot dV + \int_\Gamma (\mathbf{N}\mathbf{S})^T \mathbf{u} \cdot d\Gamma$$

$$\int_V \mathbf{S}^T (\boldsymbol{\varepsilon} - \mathbf{D}^* \mathbf{u}) dV = 0$$

$$\mathbf{e} = -\mathbf{A}_V \mathbf{q}_V + \mathbf{A}_\Gamma \mathbf{q}_\Gamma + \bar{\mathbf{e}}$$

## ELASTICITY CONDITION

$$\int_V \mathbf{S}^t (\boldsymbol{\varepsilon} - \mathbf{f} \boldsymbol{\sigma}) dV = 0$$

$$\mathbf{F} = \int_V \mathbf{S}^T \mathbf{f} \mathbf{S} dV$$

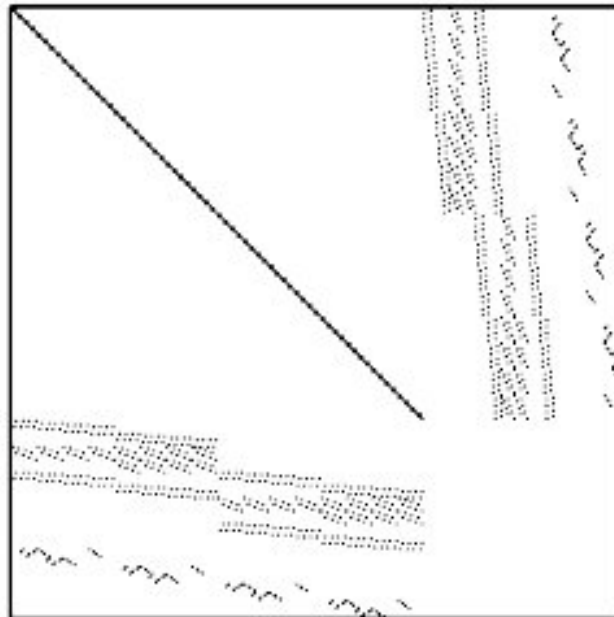
$$\mathbf{e} = \mathbf{F} \mathbf{X}$$





## GOVERNING SYSTEM

$$\begin{bmatrix} \mathbf{F} & \mathbf{A}_V & -\mathbf{A}_\Gamma \\ \mathbf{A}_V^T & \cdot & \cdot \\ -\mathbf{A}_\Gamma^T & \cdot & \cdot \end{bmatrix} \begin{bmatrix} \mathbf{x} \\ \mathbf{q}_V \\ \mathbf{q}_\Gamma \end{bmatrix} = \begin{bmatrix} \bar{\mathbf{e}} \\ -\mathbf{Q}_V \\ -\mathbf{Q}_\Gamma \end{bmatrix}$$







## 2. Why wavelets?

### Localization & Adaptability

- Modeling singularities in tension (cracks, damage, ...)

### What we look for in functions...

- Hierarchical
- Orthogonal
- Fast computation and numerical analysis

### 2.1 What work was previously done?

- The use of only scaling functions (Daubechies orthogonal wavelets)
  - Elasticity
  - Plasticity
- Manipulation of the wavelets
  - Problems with the boundaries
  - Loss of orthogonality



## 2.2 Options

- Wavelets on the Interval
- Use of other wavelet systems even if not orthogonal

## 2.3 What we did...

- Application of Daubechies orthogonal Wavelets on the Interval based on the works of:
  - A. Cohen, I. Daubechies and P. Vial;
  - V. Perrier and P. Monasse

$$\phi_{j,k}^{\text{Left}}(x) = \sum_{l=0}^{N-1} H_{k,l}^{\text{Left}} \phi_{j+1,k}^{\text{Left}}(x) + \sum_{m=N}^{N+2C} h_{k,m}^{\text{Left}} \phi_{j+1,m}^{\text{Left}}(x)$$

- $\phi_{j,k}^{\text{Left}}(x) = 2^{j/2} \phi_{0,k}^{\text{Left}}(2^j x)$
  - $\phi_{j,m}(x) = 2^{j/2} \phi(2^j x - m)$
  - $C=2k$  where  $k=0, \dots, N-1$  (Cohen)
  - $C=N-1$  (Perrier)
- Using numerical integration



### 3. Some results

#### Problems

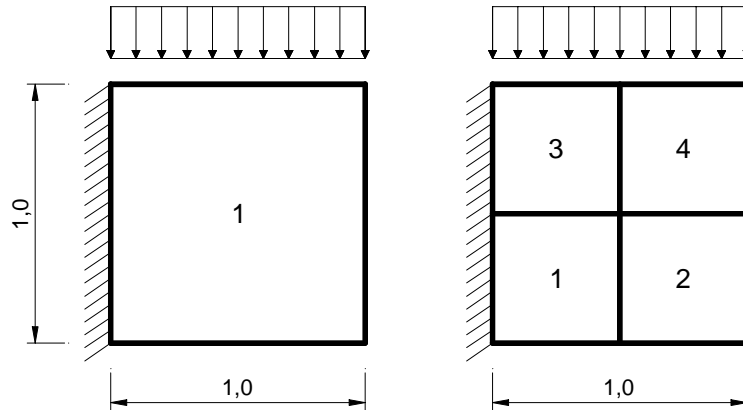
- Stretching plates
  - Short Cantilever
  - Stressed stretching plate with central crack
  - L plate

#### Approximations

- Type 1: Scaling functions + wavelets
- Type 2: Only scaling functions



- Short Cantilever ( $E = 1.0$ ,  $\nu = 0.3$ )

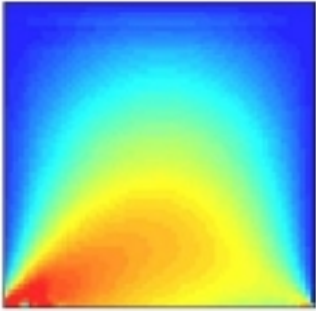
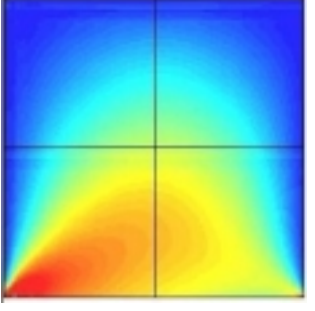
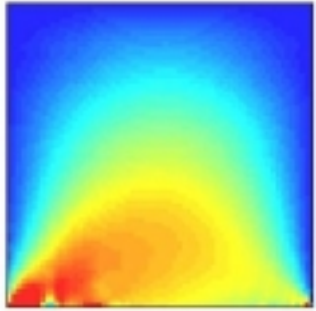
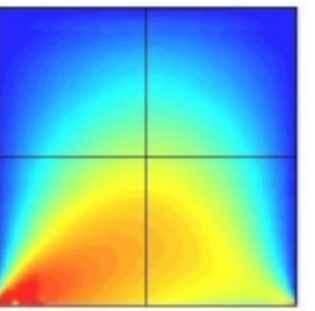
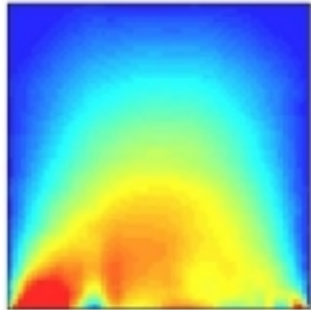
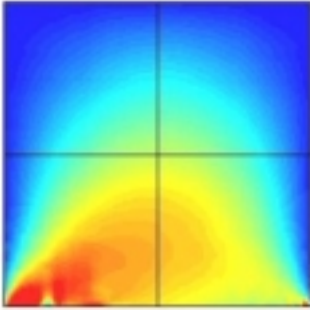


Mesh A

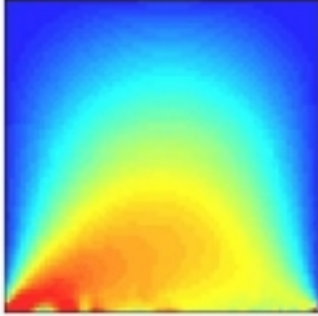
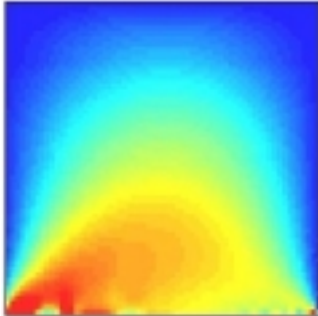
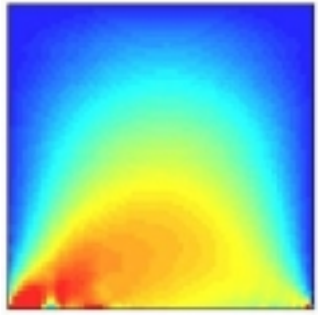
Mesh B

Problem	Model		nele	N	jo	jx	jv	jg	$\alpha_v$	$\beta$	ndf	nnz	spar
C	T 2	Cdis	1	4	-	4	3	3	768	176	<b>944</b>	48196	0.8927
C	T 2	Cdis1	1	4	-	5	4	4	3072	608	<b>3680</b>	186096	0.9727
C	T 2	Cdis4	1	5	-	5	4	4	3072	608	<b>3680</b>	286054	0.9579
C	T 2	Cdis7	1	6	-	5	4	4	3072	608	<b>3680</b>	415740	0.9388
C	T 1	Cdis	1	4	3	4	3	3	3072	608	<b>3680</b>	38016	0.9946
C	T 2	Cdis10	4	4	-	4	3	3	3072	672	<b>3744</b>	194176	0.9725
C	T 1	Cdis1	1	4	3	5	4	4	12288	2240	<b>14528</b>	246272	0.9977
C	T 2	Cdis11	4	4	-	5	4	4	12288	2368	<b>14656</b>	741232	0.9931
C	T 1	Cdis3	1	4	3	5	5	4	12288	8384	<b>20672</b>	783360	0.9963
C	T 2	Cdis12	4	4	-	6	5	5	49152	8832	<b>57984</b>	2812224	0.9983

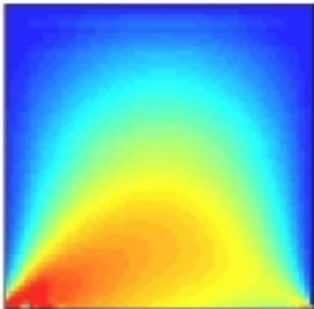
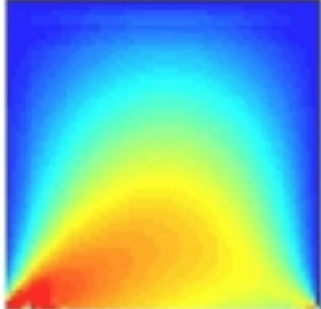
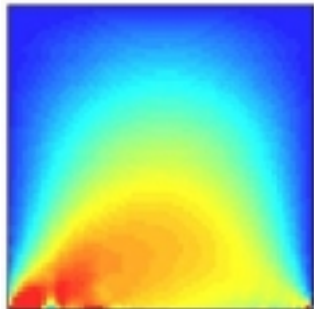
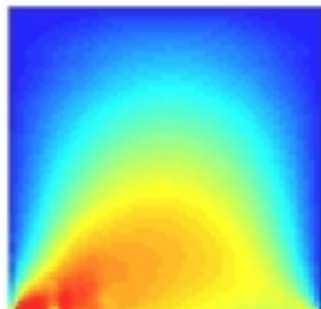
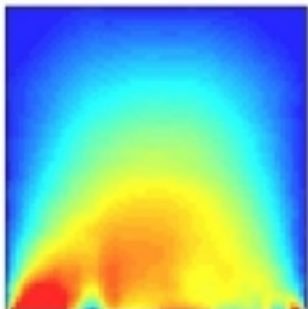
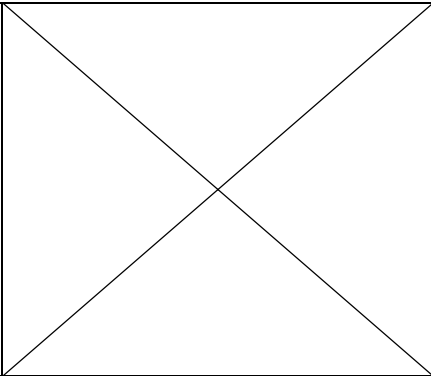


<b>N = 4</b>		T2_CDIS 2	T2_CDIS 12
	JX=6, JV=5, JG=5	 NNZ=710 084, NDF=14 528	 NNZ=2 812 224, NDF=57 984
		T2_CDIS 1	T2_CDIS 11
	JX=5, JV=4, JG=4	 NNZ=186 096, NDF=3 680	 NNZ=741 232, NDF=14 656
		T2_CDIS	T2_CDIS 10
	JX=4, JV=3, JG=3	 NNZ=48 196, NDF= 944	 NNZ=194 176, NDF= 3 744
		1 ELE	4 ELE



		T2_CDIS 7	
<b>N = 6</b>	JX=5, JV=4, JG=4		NNZ=415 740, NDF=3 680
	T2_CDIS 4		
	JX=5, JV=4, JG=4		NNZ=286 054, NDF=3 680
<b>N = 5</b>	T2_CDIS 1		
	JX=5, JV=4, JG=4		NNZ=186 096, NDF=3 680
<b>N = 4</b>		1 ELE	



<b>N = 4</b>		T2_CDIS 2		T1_CDIS 1
	JX=6, JV=5, JG=5	 NNZ=710 084, NDF=14 528	JX=5, JV=4, JG=4	 NNZ=246 272, NDF=14 528
		T2_CDIS 1		T1_CDIS
	JX=5, JV=4, JG=4	 NNZ=186 096, NDF=3 680	JX=4, JV=3, JG=3	 NNZ=38 016, NDF=13 680
	T2_CDIS			
JX=4, JV=3, JG=3	 NNZ=48 196, NDF= 944			
T 2			T 1 J0=3	



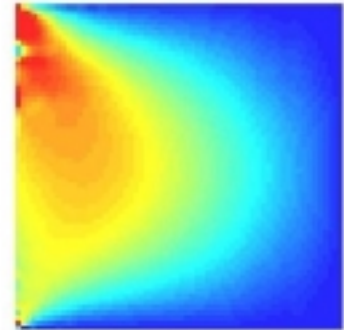
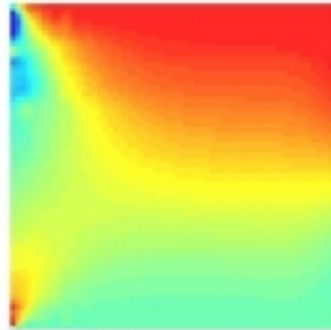
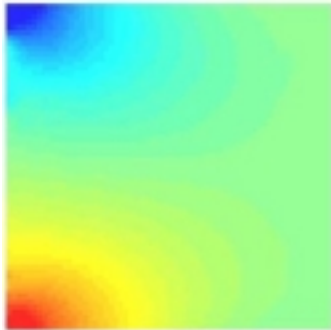


## T1\_CDIS

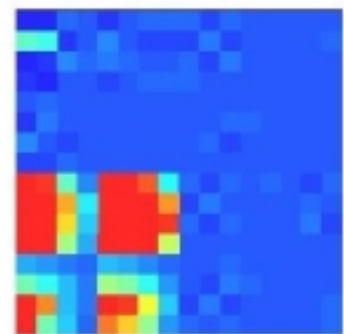
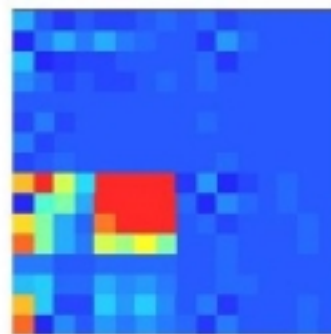
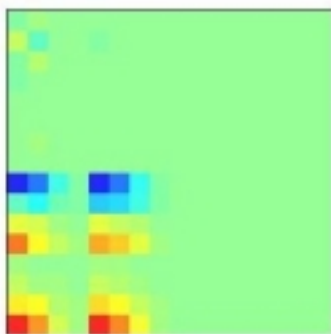
$\sigma_{xx}$

$\sigma_{yy}$

$\sigma_{xy}$



## T1\_CDIS - MULTIREOLUTION



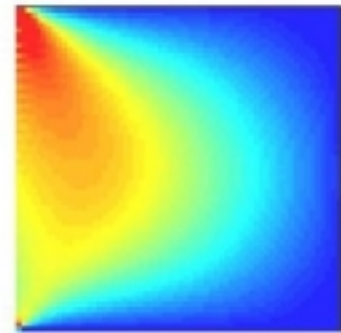
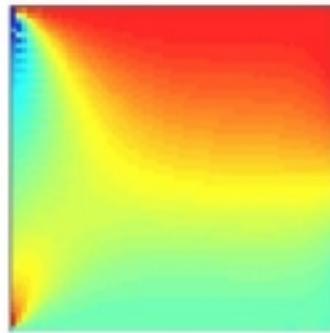
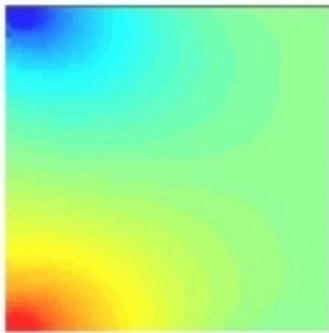


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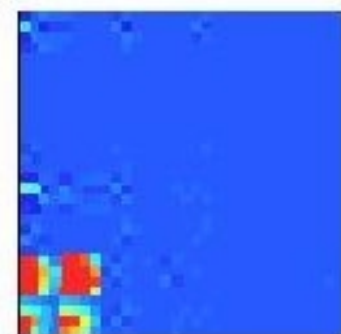
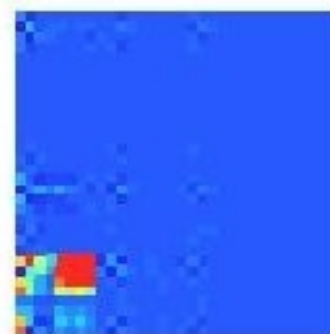
$\sigma_{xx}$

$\sigma_{yy}$

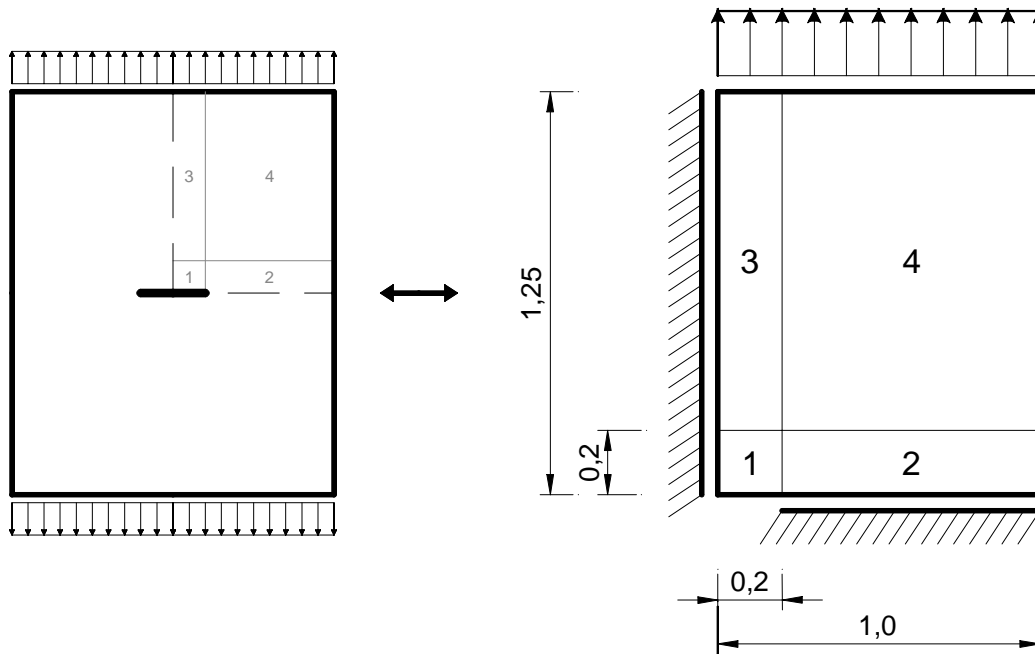
$\sigma_{xy}$



## T1\_CDIS3 - MULTIREOLUTION



- Stressed stretching plate with central crack



Problem	Model		N	jo	jx	jv	yg	$\alpha_v$	$\beta$	ndf	nnz	spar
Crack	T 2	T	4	-	5	4	4	12288	2384	<b>14672</b>	740596	0.9931
Crack	T 1	T	4	3	4	3	3	12288	2384	<b>14672</b>	153024	0.9986
Crack	T 1	T1	4	3	5	4	4	49152	8864	<b>58016</b>	987648	0.9994
Crack	T 1	T 2	4	3	5	5	4	49152	33440	<b>82592</b>	3136000	0.9990

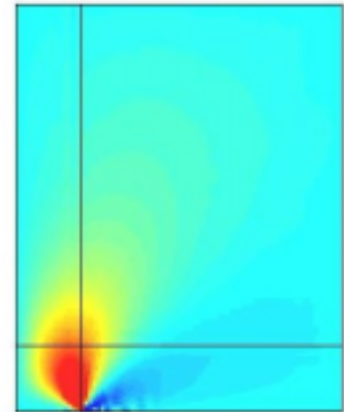
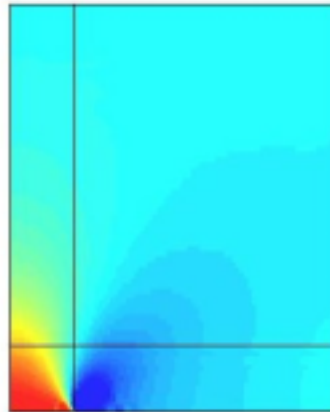
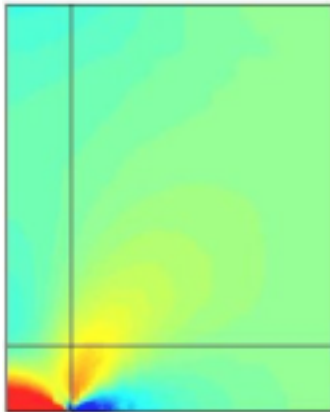


## T2 T

$\sigma_{xx}$

$\sigma_{yy}$

$\sigma_{xy}$



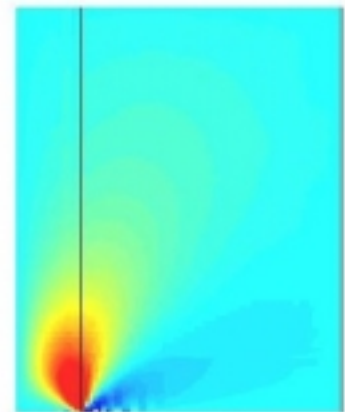
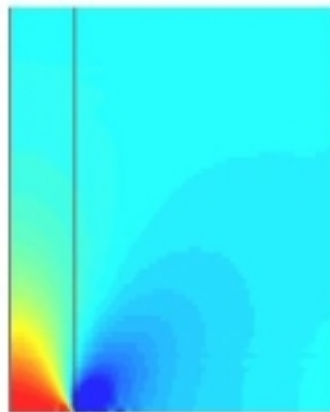
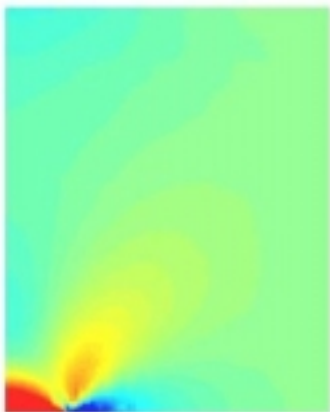


T1 T

$\sigma_{xx}$

$\sigma_{yy}$

$\sigma_{xy}$

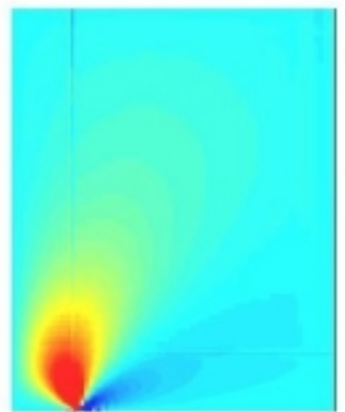
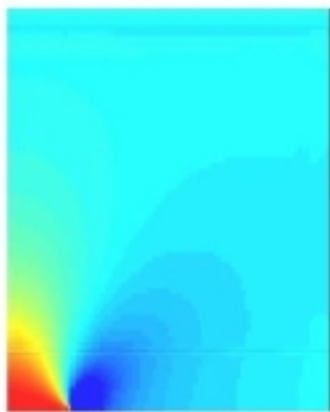
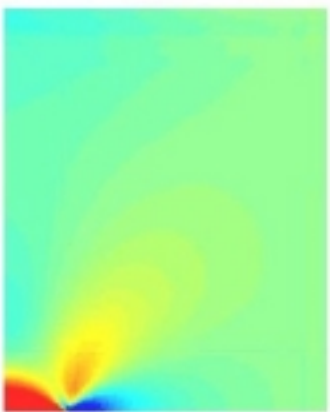


T1 T1

$\sigma_{xx}$

$\sigma_{yy}$

$\sigma_{xy}$



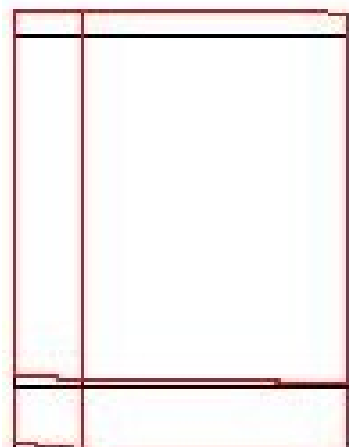
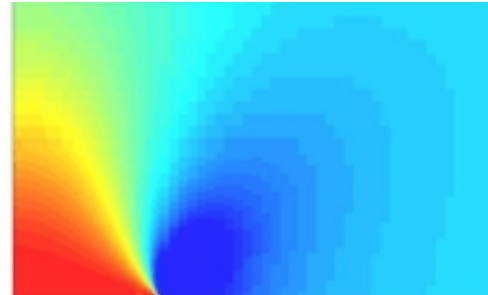
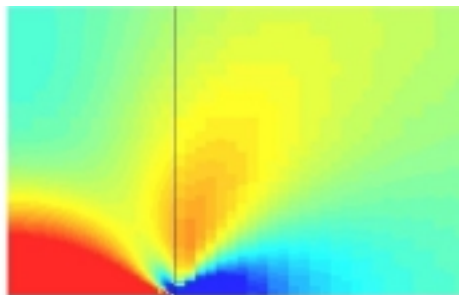
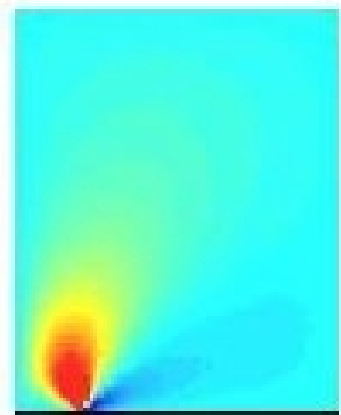
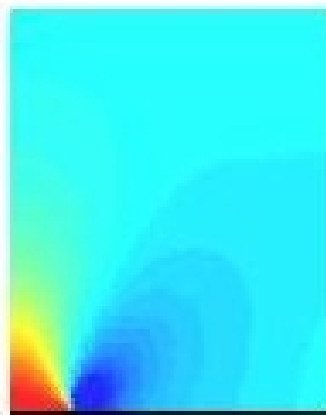
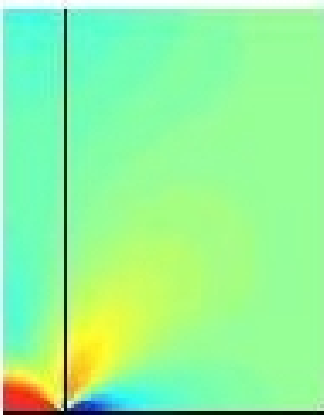


T1 T2

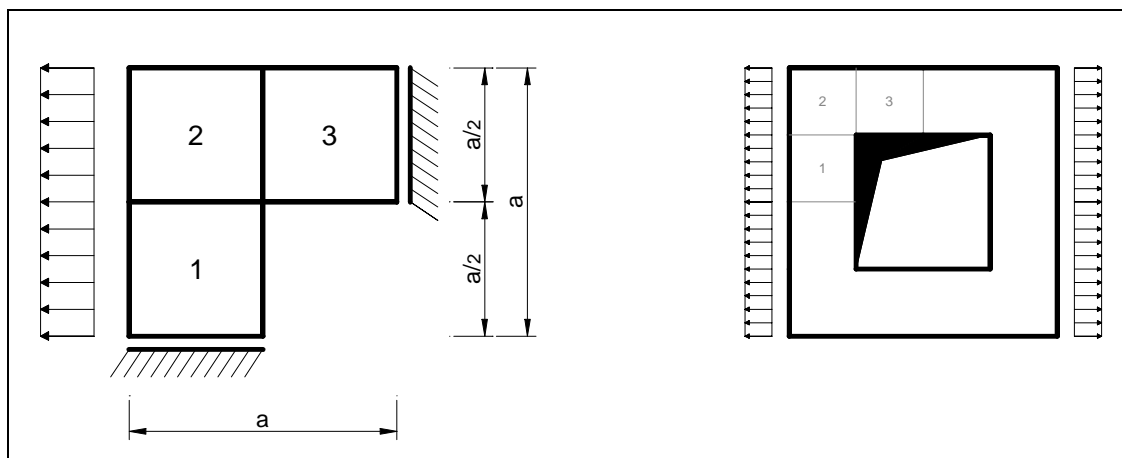
$\sigma_{xx}$

$\sigma_{yy}$

$\sigma_{xy}$



- L – stretching plate

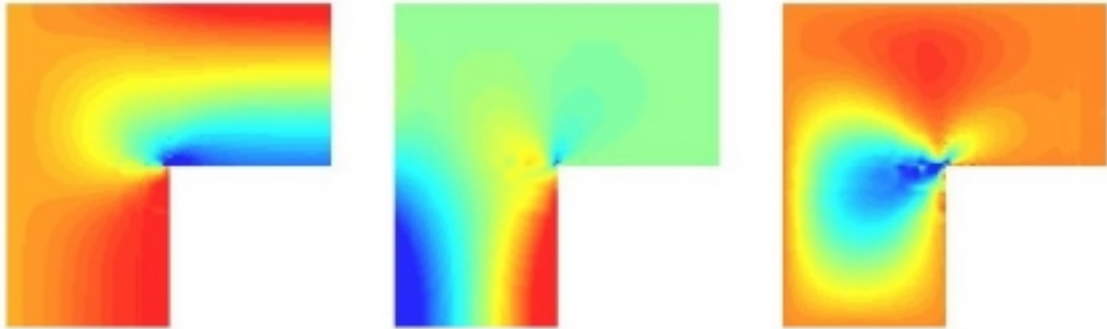


Problem	Model		N	$j_o$	$j_x$	$j_v$	$j_g$	$\alpha_v$	$\beta_v$	$\beta_\gamma$	ndf	nnz	Spar
L	T 1	L	4	3	4	3	3	9216	1536	288	<b>11040</b>	114812	0.9981
L	T 1	L 2	4	3	5	5	4	36864	24576	576	<b>62016</b>	2352128	0.9987

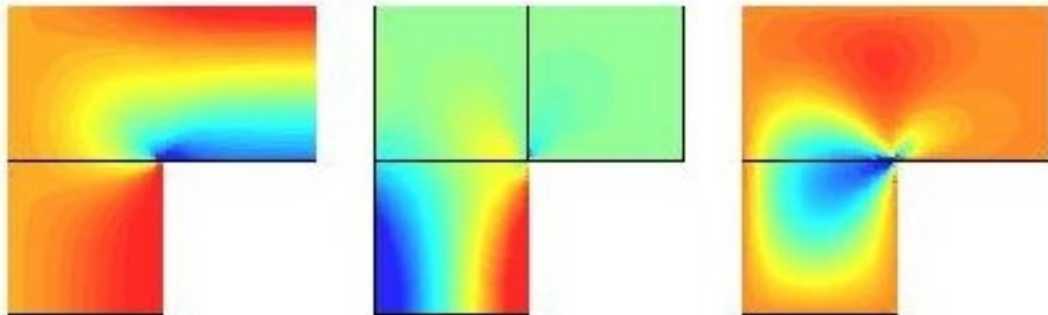




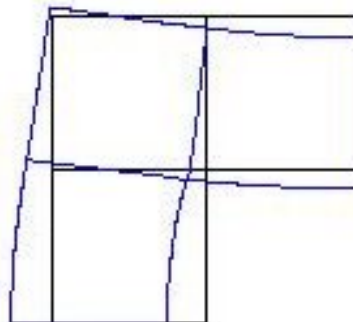
T1\_L



T1\_L2



DEFORMATION





## 4. Future work ...

- Wavelet related
  - Implementation of analytical integrations based on works of Beylkin, Dahmen and Michelli, and Perrier
  - Comparison between Cohen's and Perrier's Wavelets on the Interval
  - Study of other wavelet systems on the interval
    - Orthogonal (Interacting boundary wavelet)
    - Non orthogonal (bi-orthogonal, ...)
  - Implementation of adaptive schemes
  - Physical non-linear analysis (Elastoplasticity, fracture and damage mechanics)
  - 3D models